# Global Positioning System (GPS) Standard Positioning Service (SPS) Performance Analysis Report

**Submitted To** 

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Submitted by

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### **Executive Summary**

The GPS Product Team has tasked the Navigation Branch at the William J. Hughes Technical Center to document the Global Positioning System (GPS) Standard Positioning Service (SPS) performance in quarterly GPS Performance Analysis (PAN) Reports. The report contains the analysis performed on data collected at twenty-eight Wide Area Augmentation System (WAAS) Reference Stations. This analysis verifies the GPS SPS performance as compared to the performance parameters stated in the SPS Specification (September 2008).

This report, Report #74, includes data collected from 1 April through 30 June 2011. The next quarterly report will be issued October 31, 2011.

Analysis of this data includes the following standards and categories: PDOP Availability, NANU Summary and Evaluation, Service Availability, Position and Range Accuracy and Solar Storm Effects on GPS SPS performance.

PDOP availability is based on Position Dilution of Precision (PDOP). Utilizing the weekly almanac posted on the US Coast Guard navigation web site, the coverage for every 5° grid point between 180W to 180E and 80S and 80N was calculated for every minute over a 24-hour period for each of the weeks covered in the reporting period. For this reporting period, the global availability based on PDOP less than six for CONUS was 100%.

NANU summary and evaluation was achieved by reviewing the "Notice: Advisory to Navstar Users" (NANU) reports issued between 1 April and 30 June 2011. Using this data, we compute a set of statistics that give a relative idea of constellation health for both the current and combined history of past quarters. A total of seven outages were reported in the NANU's this quarter. Five outages were scheduled while two were unscheduled outages.

The quarterly service availability standard was verified using 24-hour position accuracy values computed from data collected at one-second intervals. All of the sites achieved a 100% availability, which exceeds the SPS "average location" value of 99% and the "worst-case location" value of 90%.

Calculating the 24-hour 95% horizontal and vertical position error values verified the accuracy standards. The User Range Error standard was verified for each satellite from 24-hour accuracy values computed using data collected at the following six sites: Boston, Honolulu, Los Angeles, Miami, San Juan and Juneau. This data was also collected in one-second samples. All sites achieved 100% reliability, meeting the SPS specification. The maximum range error recorded was 22.116 meters on Satellite PRN 22. The SPS specification states that the range error should never exceed 30 meters for less than 99.79% of the day for a worst-case point and 99.94% globally. The maximum RMS range error value of 2.569 recorded on satellite PRN 22. The SPS specification states that RMS URE cannot exceed 6 meters in any 24-hour interval.

Geomagnetic storms had little to no effect on GPS performance this quarter. All sites met all GPS Standard Positioning Service (SPS) specifications on those days with the most significant solar activity.

The IGS is a voluntary federation of many worldwide agencies that pool resources and permanent GNSS station data to generate precise GNSS products. During the evaluation period, the maximum 95% horizontal and vertical SPS errors were 6.72 meters at Maspalomas, Spain and 7.94 meters at Puerto Ayora, Ecuador respectively.

From the analysis performed on data collected between 1 April and 30 June 2011, the GPS performance met all SPS requirements that were evaluated. There were no significant problems to report for the duration of the quarter.

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## **1** Introduction

### 1.1 Objective of GPS SPS Performance Analysis Report

In 1993, the FAA began monitoring and analyzing Global Positioning System (GPS) Standard Positioning Service (SPS) performance data. At present, the FAA has approved GPS and WAAS for IFR operations and is developing Local Area Augmentation (LAAS), which is an additional GPS augmentation system. In order to ensure the safe and effective use of GPS and its augmentation systems within the NAS, it is critical that characteristics of GPS performance as well as specific causes for service outages be monitored and understood. To accomplish this objective, GPS SPS performance data is documented in a quarterly GPS Analysis report. This report contains data collected at the following twenty-eight WAAS reference station locations:

- Bethel, A K
- Billings, MT
- Fairbanks, AK
- Cold Bay, AK
- Kotzebue, AK
- Juneau, AK
- Albuquerque, NM
- Anchorage, AK
- Boston, MA
- Washington, D.C.
- Honolulu, HI
- Houston, TX
- Kansas city, KS
- Los Angeles, CA
- Salt Lake City, UT
- Miami, FL
- Minneapolis, MI
- Oakland, CA
- Cleveland, OH
- Seattle, WA
- San Juan, PR
- Atlanta, GA
- Barrow, AK
- Merida, Mexico
- Gander, Canada
- Tapachula, Mexico
- San Jose Del Cabo, Mexico
- Iqaluit, Canada

The analysis of the data is divided into the four performance categories stated in the Standard Positioning Service Performance Specification (September 2008). These categories are:

- PDOP Availability Standard
- Service Availability Standard
- Service Reliability Standard
- Positioning, Ranging and Timing Accuracy Standard

The results were then compared to the performance parameters stated in the SPS.

### 1.2 Report Overview

Section 2 of this report summarizes the results obtained from the coverage calculation program developed by the GPS test team. The SPS coverage area program uses the GPS satellite almanacs to compute each satellite position as a function of time for a selected day of the week. This program establishes a 5-degree grid between 180 degrees east and 180 degrees west, and from 80 degrees north and 80 degrees south. The program then computes the PDOP at each grid point (1485 total grid points) every minute for the entire day and stores the results. After the PDOP's have been saved the 99.99% index of 1-minute PDOP at each grid point is determined and plotted as contour lines (Figure 2-1). The program also saves the number of satellites used in PDOP calculation at each grid point for analysis.

Section 3 summarizes the GPS constellation performance by providing the "Notice: Advisory to Navstar Users" (NANU) messages to calculate the total time of forecasted and actual satellite outages. This section also evaluates the Service Availability Standard using 24-hour 95% horizontal and vertical position accuracy values.

Section 4 summarizes service reliability performance. It will be reported at the end of the first year of this analysis because the SPS standard is based on a measurement interval of one year. Data for the quarter is provided for completeness.

Section 5 provides the position accuracies based on data collected on a daily basis at one-second intervals. This section also provides the statistics on the range error, range error rate and range acceleration error for each satellite. The overall average, maximum, minimum and standard deviations of the range rates and accelerations are tabulated for each satellite.

In Section 6, the data collected during solar storms is analyzed to determine the effects, if any, of GPS SPS performance.

Section 7 provides an analysis of GPS-SPS accuracy performance from a selection of high rate IGS stations around the world.

Section 8 provides a summary of GPS Test NOTAMs.

Section 9 provides four appendices to summarize the data found in this report and provide further information.

Appendix A provides a summary of all the results as compared to the SPS specification.

Appendix B provides the geomagnetic data used for Section 6.

Appendix C provides a PAN Problem Report.

Appendix D provides a glossary of terms used in this PAN report. This glossary was obtained directly from the GPS SPS specification document (October 2001).

### 1.3 Summary of Performance Requirements and Metrics

Table 1-1 over the next four pages lists the performance parameters from the SPS and identifies those parameters verified in this report.

Per-Satellite Coverage	Conditions and Constraints	Evaluated in This Report
Terrestrial Service Volume: 100% Coverage	• For any health or marginal SPS SIS	Future Report
Space Service Volume: No Coverage Performance		
Specified		
Constellation Coverage	Conditions and Constraints	
Terrestrial Service Volume: 100% Coverage	• For any healthy or marginal SPS SIS	Future Report
Space Service Volume: No Coverage Performance		Report
Specified		
User Range Error	Conditions and Constraints	
Accuracy		
Single Frequency C/A-Code	• For any healthy SPS SIS	
$\bullet \le 7.8 \text{m} 9\%\%$ Global	• Neglecting single-frequency ionospheric delay model	
Average URE during normal	errors	
operations over All AODs	• Including group delay time correction (T <sub>GD</sub> ) errors at	
$\bullet \le 6.0 \text{m} 95\%$ Global	L1	$\checkmark$
Average URE during	• Including inter-signal bias (P(Y)-code to C/A-code)	
operations at Zero AOD	errors at L1	
• $\leq 12.8 \text{m} 95\%$ Global		
Average URE during normal operations at Any AOD		
Single Frequency C/A-Code	• For any healthy SPS SIS.	
Single Hequency GIT code	<ul> <li>Neglecting single-frequency ionospheric delay model</li> </ul>	
• $\leq 30m 99.94\%$ Global	errors	
Average URE during normal	• Including group delay time correction (T <sub>GD</sub> ) errors at	
operations	L1	/
	• Including inter-signal bias (P(Y)-code to C/A-code)	$\checkmark$
• $\leq$ 30m 99.79% Worst	errors at L1	
Case single point average	• Standard based on measurement interval of one year; average of daily values within service volume	
during normal operations.	<ul> <li>Standard based on 3 service failures per year, lasting</li> </ul>	
	no more than 6 hours each	
User Range Rate Error Accuracy	Conditions and Constraints	
Single-Frequency C/A-	• For any healthy SPS SIS	
Code:	<ul> <li>Neglecting all perceived pseudorange rate errors</li> </ul>	
	attributable to pseudorange step changes caused by	
• $\leq 6 \text{ mm/sec } 95\%$ Global	NA V message data cutovers	
Average URRE over any 3-	• Neglecting single-frequency ionospheric delay model	•
second interval during	errors	
normal operations at Any		
AOD		

#### Table 1-1 SPS SIS Performance Requirements Standards

User Range Acceleration Error Accuracy	Conditions and Constraints	Evaluated in This Report
Single-Frequency C/A- Code: • ≤ 2 mm/sec <sup>2</sup> 95% Global average URAE over any 3- second interval during normal operations at Any AOD	<ul> <li>For any healthy SPS SIS</li> <li>Neglecting all perceived pseudorange rate errors attributable to pseudorange step changes caused by NA V message data cutovers</li> <li>Neglecting single-frequency ionospheric delay model errors</li> </ul>	$\checkmark$
Coor dinated Universal		
Time Offset Error		
Accuracy • ≤ 40 nanoseconds 95% Global average UTCOE during normal operations at Any AOD.	For any healthy SPS SIS	$\checkmark$
Instantaneous URE	Conditions and Constraints	
Integrity Single-Frequency C/A- Code: • $\leq 1 \times 10^{-5}$ Probability over any hour of the SPS SIS Instantaneous URE exceeding the NTE tolerance without a timely alert during normal operations.	<ul> <li>For any healthy SPS SIS</li> <li>SPS SIS URE NTE tolerance defined to be ±4.42 times the upper bound on the URA value corresponding to the URA index "N" currently broadcast by the satellite.</li> <li>Given that the maximum SPS SIS instantaneous URE did not exceed the NTE tolerance at the start of the hour</li> <li>Worst case for delayed alert is 6 hours.</li> <li>Neglecting singe-frequency ionospheric delay model errors</li> </ul>	Future Report
Instantaneous UTCOE Integrity	Conditions and Constraints	
Single-Frequency C/A- Code: • ≤ 1x10 <sup>-5</sup> Probability over any hour of the SPS SIS Instantaneous UTCOE exceeding the NTE tolerance without a time ly alert during normal operations.	<ul> <li>For any healthy SPS SIS</li> <li>SPS SIS URE NTE tolerance defined</li> </ul>	Future Report
Unscheduled Failure	Conditions and Constraints	
Interruption Continuity Unscheduled Failure Interruptions: • ≥ 0.9998 Probability over any hour of not losing the SPS SIS availability from a slot due to unscheduled interruption	<ul> <li>Calculated as an average over all slots in the 24-slot constellation, normalized annually</li> <li>Given that the SPS SIS is available from the slot at the start of the hour</li> </ul>	Future Report

Status and Problem Reporting	Conditions and Constraints	Evaluated in This Report
Scheduled event affecting service • Appropriate NANU issued to the Coast Guard and the FAA at least 48 hours prior to the event	• For any SPS SIS	$\checkmark$
Unscheduled outage or problem affecting service • Appropriate NANU issued to the Coast Guard and the FAA as soon as possible after the event	• For any SPS SIS	$\checkmark$
Per-Slot Availability	Conditions and Constraints	
<ul> <li>≥ 0.957 Probability that a slot in the baseline 24-slot configuration will be occupied by a satellite broadcasting a healthy SPS SIS</li> <li>≥ 0.957 Probability that a slot in the expanded configuration will be occupied by a pair of satellites each broadcasting a health SPS SIS</li> </ul>	<ul> <li>Calculated as an average over all slots in the 24-slot constellation, normalized annually</li> <li>Applies to satellites broadcasting a healthy SPS SIS that also satisfy the other performance standards in the SPS performance standard.</li> </ul>	Quarter 4 Reports Only
Constellation Availability	Conditions and Constraints	
<ul> <li>≥ 0.98 Probability that at least 21 slots out of the 24 will be occupied either by a satellite broadcasting a healthy SPS SIS in the baseline 24-slot configuration or by a pair of satellites each broadcasting a healthy SPS SIS in the expanded slot configuration</li> <li>≥ 0.99999 Probability that at least 20 slots out of the 24 will be occupied either by a satellite broadcasting a healthy SPS SIS in the baseline 24-slot configuration or by a pair of satellite broadcasting a healthy SPS SIS in the expanded slot configuration or by a pair of satellite broadcasting a healthy SPS SIS in the baseline 24-slot configuration or by a pair of satellites each broadcasting a healthy SPS SIS in the expanded slot configuration</li> </ul>	<ul> <li>Calculated as an average over all slots in the 24-slot constellation, normalized annually.</li> <li>Applies to satellites broadcasting a healthy SPS SIS that also satisfies the other performance standards in the SPS performance standard.</li> </ul>	Quarter 4 Reports Only
Operational Satellite Count	Conditions and Constraints	
• ≥ 0.95 Probability that the constellation will have at least 24 operational satellites regardless of whether those operational satellites are located in slots or not	• Applies to the total number of operational satellites in the constellation (averaged over any day); where any satellite which appears in the transmitted navigation message almanac is defined to be an operation satellite regardless of whether that satellite is currently broadcasting a healthy SPS SIS or not and regardless of whether the broadcast SPS SIS also satisfies the other performance standards in the SPS performance standard or not.	$\checkmark$

PDOP Availability	Conditions and Constraints	Evaluated in This Report
<ul> <li>≥ 98% global PDOP of 6 or less</li> <li>≥ 88% worst site PDOP of 6 or less</li> </ul>	• Defined for a position/time solution meeting the representative user conditions and operating within the service volume over any 24-hour interval	
Service Availability	Conditions and Constraints	
<ul> <li>≥ 99% Horizontal Service Availability, average location</li> <li>≥ 99% Vertical Service Availability, average location</li> </ul>	<ul> <li>17m Horizontal (SIS only) 95% threshold</li> <li>37m Vertical (SIS only) 95% threshold</li> <li>Defined for a position/time solution meeting the representative user conditions and operating within the service volume over any 24-hour interval.</li> </ul>	$\checkmark$
<ul> <li>≥ 90% Horizontal Service Availability, worst- case location</li> <li>≥ 90% Vertical Service Availability, worst-case location</li> </ul>	<ul> <li>17m Horizontal (SIS only) 95% threshold</li> <li>37m Vertical (SIS only) 95% threshold</li> <li>Defined for a position/time solution meeting the representative user conditions and operating within the service volume over any 24-hour interval.</li> </ul>	$\checkmark$
Position/Time Accuracy	Conditions and Constraints	
Global Average Position Domain Accuracy • ≤ 9m 95% Horizontal Error • ≤ 15m 95% Vertical Error	<ul> <li>Defined for a position/time solution meeting the representative user conditions</li> <li>Standard based on a measurement interval of 24 hours averaged over all points in the service volume.</li> </ul>	
Worst Site Position Domain Accuracy • ≤ 17m 95% Horizontal Error • ≤ 37m 95% Vertical Error	<ul> <li>Defined for a position/time solution meeting the representative user conditions</li> <li>Standard based on a measurement interval of 24 hours averaged over all points in the service volume.</li> </ul>	$\checkmark$
Time Transfer Domain Accuracy • ≤ 40 nanoseconds time transfer error 95% of time (SIS only)	<ul> <li>Defined for a time transfer solution meeting the representative user conditions</li> <li>Standard based on a measurement interval of 24 hours averaged over all points in the service volume.</li> </ul>	$\checkmark$

# 2 PDOP Availability Standard

**PDOP Availability**: The percentage of time over any 24-hour interval that the PDOP value is less than or equal to its threshold for any point within the service volume.

**Dilution of Precision (DOP)**: The magnifying effect on GPS position error induced by mapping GPS range errors into position within the specified coordinate system through the geometry of the position solution. The DOP varies as a function of satellite positions relative to user position. The DOP may be represented in any user local coordinate desired. Examples are HDOP for local horizontal, VDOP for local vertical, PDOP for all three coordinates, and TDOP for time.

PDOP Availability Standard	Conditions and Constraints
≥ 98% global PDOP of 6 or less ≥ 88% worst site PDOP of 6 or less	• Defined for a position/time solution meeting the representative user conditions and operating within the service volume over any 24-hour interval

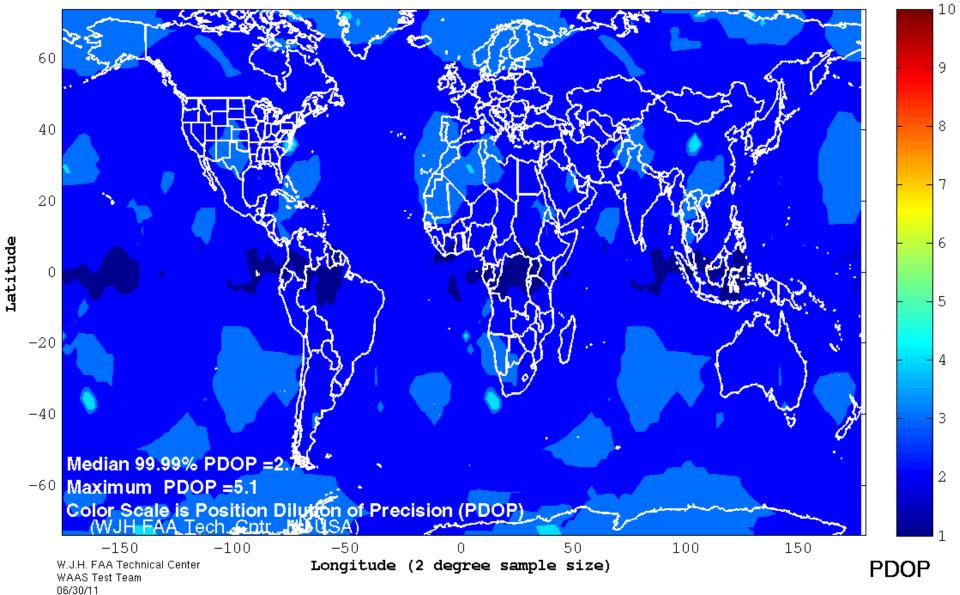
Almanacs for GPS weeks used for this coverage portion of the report were obtained from the Coast Guard web site (www.navcen.uscg.mil). Using these almanacs, an SPS coverage area program developed by the GPS test team was used to calculate the PDOP at every 5° point between longitudes of 180W to 180E and 80S and 80N at one-minute intervals. This gives a total of 1440 samples for each of the 2376 grid points in the coverage area. Table 2-1 provides the global averages and worst-case availability over a 24-hour period for each week. Table 2-1 also gives the global 99.9% PDOP value for each of the thirteen GPS Weeks. The PDOP was 3.160 or better 99.9% of the time for each of the 24-hour intervals.

Figure 2-1 is a contour plot of PDOP values over the entire globe. Inside each contour area, the PDOP value is greater than or equal to the contour value shown in the legend for that color line. That areas' value is also less than the next higher contour value, unless another contour line lies within the current area. A single "DOP hole" where the PDOP value is greater than 6 was evaluated for satellite visibility for one 24-hour interval from the week shaded in Table 2-1. The histogram in figure 2-2 shows the satellite visibility at the DOP hole position for the 24 hour interval in question.

The GPS coverage performance evaluated met the specifications stated in the SPS.

Date Range of Week	Global 99.9% PDOP Value	Global Average (Spec:≥98%)	Worst-Case Point (Spec: ≥ 88%)
3 – 9 April	2.949	100%	100%
10 – 16 April	2.937	100%	100%
17 – 23 April	2.948	100%	100%
24 – 30 April	2.900	100%	100%
1 – 7 May	2.926	100%	100%
8–14 May	2.946	100%	100%
15 – 21 May	3.123	100%	100%
22 – 28 May	3.126	100%	100%
29 May – 4 June	3.130	100%	100%
5 – 11 June	3.134	100%	100%
12 – 18 June	3.140	100%	100%
19 – 25 June	3.147	100%	100%
26 June – 2 July	3.160	100%	100%

#### Table 2-1 PDOP A vailability Statistics



# 06/30/11 World GPS Maximum PDOP

13

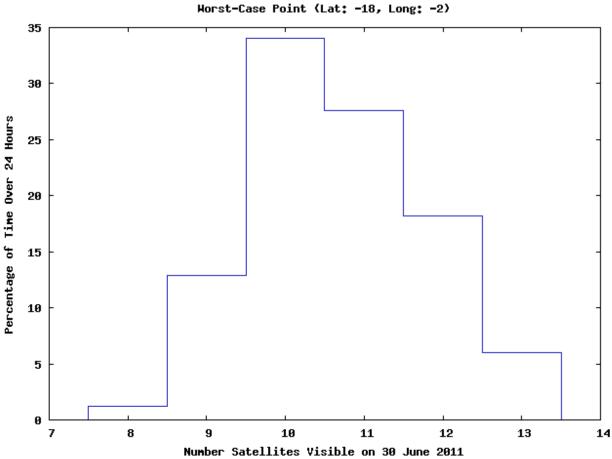


Figure 2-2 Satellite Visibility Profile for Worst-Case Point

July 31, 2011

# **3** NANU Summary and Evaluation

**NANU:** <u>N</u>otice <u>A</u>dvisory to <u>N</u>AVSTAR <u>U</u>sers – A periodic bulletin alerting users to changes in the satellite system performance.

Status and Problem Reporting	Conditions and Constraints
<ul> <li>Scheduled event affecting service</li> <li>Appropriate NANU issued to the Coast Guard and the FAA at least 48 hours prior to the event</li> </ul>	• For any SPS SIS
<ul><li>Unscheduled outage or problem affecting service</li><li>Appropriate NANU issued to the Coast Guard and the FAA as soon as possible after the event</li></ul>	• For any SPS SIS

### 3.1 Satellite Outages from NANU Reports

Satellite availability performance was analyzed based on published "Notice: Advisory to Navstar Users" messages (NANU's). During this reporting period, 1 April through 30 June 2011, there were a total of seven reported outages. Five of these outages were maintenance activities and were reported in advance while two were unscheduled outages. A complete listing of outage NANU's for the reporting period is provided in Table 3-1. A complete listing of the forecasted outage NANU's for the reporting period can be found in Table 3-2. Canceled outage NANU's (if any) are provided in Table 3-3. The minimum duration a scheduled outage was forecasted ahead of time was 132.883 hours, which exceeded the 48-hour requirement. The maximum response time for a NANU issued for an unscheduled outage was 2.10 hours.

NANU#	PRN	ТҮРЕ	Start Date	Start Ti me	End Date	End Time	Total Unscheduled	Total Scheduled	Total
2011028	18	FCSTSUMM	Apr-05	15:26	Apr-05	21:18		5.87	5.87
2011030	26	UNUSABLE	Apr-10	17:12	Apr-12	0:43	31.52		31.52
2011031	6	FCSTSUMM	Apr-12	1:35	Apr-12	5:06		3.52	3.52
2011033	29	FCSTSUMM	Apr-21	7:14	Apr-21	12:29		5.25	5.25
2011037	11	FCSTSUMM	May-03	17:33	May-04	0:00		6.45	6.45
2011038	23	FCSTSUMM	May-05	15:51	May-05	21:27		5.60	5.60
2011041	10	UNUSABLE	May-07	14:45	May-10	19:20	76.58		76.58
	Totals of Unscheduled, Scheduled & Total Downtime108.1026.69134.74								

#### Table 3-1 NANUs Affecting Satellite Availability

### **GENERAL NANUs**

2011026 (2011-Apr-05) - Testing requires the non-operational M-code signal be turned off for an indeterminate period on PRN25 (SVN62).

2011034 (2011-Apr-21) - Signal mitigation testing is on-going on PRN1 (SVN49). The satellite will remain unhealthy until further notice.

2011039 (2011-May-06) - Decommission of PRN1 (SVN49).

2011043 (2011-May-31) - Resume transmitting L-band signal on PRN1 (SVN35). The satellite will not be included in the broadcast almanac.

2011044 (2011-Jun-20) - Installation of new NANU software

2011045 (2011-Jun-22) - Completed Installation of new NANU software

NANU#	PRN	Туре	Start Date	Start Ti me	End Date	End Time	Total	Comments
2011025	18	FCSTDV	5-Apr	15:20	6-Apr	3:20	12	2011028
2011027	6	FCSTMX	11-Apr	21:00	12-Apr	9:00	12	2011031
2011029	26	UNUSUFN	10-Apr	17:12				2011030
2011032	29	FCSTDV	21-Apr	6:45	21-Apr	18:45	12	2011033
2011035	11	FCSTDV	3-May	17:00	4-May	5:00	12	2011037
2011036	23	FCSTDV	5-May	15:15	6-May	3:15	12	2011038
2011040	10	UNUSUFN	7-May	14:45				2011041
2011042	30	UNUSUFN	13-May	20:55				N/A
	Total Forecasted Downtime							

#### Table 3-2 NANUs Forecasted to Affect Satellite Availability

#### Table 3-3 Cancelled NANUs

NANU#	PRN	Туре	Start Date	Start Time	Comments
None					

Satellite Reliability, Maintainability, and Availability (RMA) data is being collected based on published "Notice: Advisory to Navstar Users" messages (NANU's). This data has been summarized in Table 3-4. The "Total Satellite Observed MTTR" was calculated by taking the average downtime of all satellite outage occurrences. Scheduled downtime was forecasted in advance via NANU's. All other downtime reported via NANU was considered unscheduled. The "Percent Operational" was calculated based on the ratio of total actual operating hours to total available operating hours for every satellite.

#### Table 3-4 GPS Satellite Maintenance Statistics

Satellite Reliability/Maintainability/Availability (RMA) Parameter	1-Apr-11 30-Jun-11	1-Jan-00 30-Jun-11
Total Forecast Downtime (hrs):	60.00	8,647.72
Total Actual Downtime (hrs):	134.79	36,886.13
Total Actual Scheduled Downtime (hrs):	36.69	4,693.17
Total Actual Unscheduled Downtime (hrs):	108.10	32,192.96
Total Satellite Observed MTTR (hrs):	19.26	53.77
Scheduled Satellite Observed MTTR (hrs):	5.34	8.82
Unscheduled Satellite Observed MTTR (hrs):	54.05	209.05
# Total Satellite Outages:	7	686
# Scheduled Satellite Outages:	5	532
# Unscheduled Satellite Outages:	2	154
Percent Operational Scheduled Downtime:	99.96	99.85
Percent Operational All Downtime:	99.80	98.82

### 3.2 Service Availability Standard

**Service Availability:** The percentage of time over any 24-hour interval that the predicted 95% position error is less than the threshold at any given point within the service volume.

• Horizontal Service Availability: The percentage of time over any 24-hour interval that the predicted 95% horizontal error is less than its threshold for any point within the service volume.

• Vertical Service Availability: The percentage of time over any 24-hour interval that the predicted 95% vertical error is less than its threshold for any point within the service volume.

Service Availability Standard	Conditions and Constraints
• $\geq$ 99% Horizontal Service Availability, average	• 17m Horizontal (SIS only) 95% threshold
location	• 37m Vertical (SIS only) 95% threshold
	• Defined for a position/time solution meeting the
• $\geq$ 99% Vertical Service Availability, average location	representative user conditions and operating within the
	service volume over any 24-hour interval.
• $\geq$ 90% Horizontal Service Availability, worst-case	• 17m Horizontal (SIS only) 95% threshold
location	• 37m Vertical (SIS only) 95% threshold
	• Defined for a position/time solution meeting the
• $\geq$ 90% Vertical Service Availability, worst-case	representative user conditions and operating within the
location	service volume over any 24-hour interval.

To verify availability, the data collected from receivers at the twenty-eight WAAS sites was reduced to calculate 24hour accuracy information and reported in Table 3-5. The data was collected at one-second intervals between 1 April and 30 June 2011.

Site	Total Number of Seconds of SPS Monitoring	Instances of 24-hour Threshold Failures	Quarters Service Availability %
Al bu quer que	7858683	0	100%
Anchorage	7858891	0	100%
Atlanta	7859056	0	100%
Barrow	7848226	0	100%
Bethel	7851394	0	100%
Billings	7858895	0	100%
Boston	7859046	0	100%
Cleveland	7859058	0	100%
Cold Bay	7852473	0	100%
Fairbanks	7686418	0	100%
Gander	7849930	0	100%
Honolulu	7858754	0	100%
Houston	7859019	0	100%
Iqaluit	7843791	0	100%
Juneau	7837219	0	100%
Kansas City	7858724	0	100%
Kotzebue	7853977	0	100%
Los Angeles	7858965	0	100%
Merida	7843715	0	100%
Miami	7857168	0	100%
Minneapolis	7858941	0	100%
Oakland	7857242	0	100%
Salt Lake City	7858962	0	100%
San Jose Del Cabo	7850341	0	100%
San Juan	7858988	0	100%
Seattle	7853863	0	100%
Tapachula	4754450	0	100%
Washington, DC	7857564	0	100%
G	obal Average over Reporting Per	iod = 100% (SPS Spec. > 95	.87%)

#### Table 3-5 Accuracies Exceeding Threshold Statistics

# 4 Service Reliability Standard

**Service Reliability:** The percentage of time over a specific time interval that the instantaneous SIS SPS URE is maintained within a specified reliability threshold at any given point within the service volume, for all healthy GPS satellites.

User Range Error Accuracy	Conditions and Constraints
	• For any healthy SPS SIS.
Single Frequency C/A-Code	• Neglecting single-frequency ionospheric delay model
	errors
• ≤ 30m 99.94% Global Average URE during normal	• Including group delay time correction (T <sub>GD</sub> ) errors at
operations	L1
	• Including inter-signal bias (P(Y)-code to C/A-code)
• $\leq$ 30m 99.79% Worst Case single point average	errors at L1
during normal operations.	• Standard based on measurement interval of one year;
	average of daily values within service volume
	• Standard based on 3 service failures per year, lasting
	no more than 6 hours each

Table 4-1 shows a comparison to the service reliability standard for range data collected at a set of six receivers across North America. Although the specification calls for yearly evaluations, we will be evaluating this SPS requirement at quarterly intervals. Additional range analysis results can be found in table 5-2. The maximum User Range Error recorded this quarter was 22.116 meters on satellite PRN 22.

#### Table 4-0-1 User Range Error Accuracy

Date Range of Data Collection	Site	Number of Samples This Quarter	Number of Samples where SPS URE > 30m NTE	Percentage
1 Apr – 30 Jun 2011	Boston	66,153,335	0	100%
1 Apr – 30 Jun 2011	Honolulu	68,800,556	0	100%
1 Apr – 30 Jun 2011	Los Angeles	68,308,238	0	100%
1 Apr – 30 Jun 2011	Miami	66,140,232	0	100%
1 Apr – 30 Jun 2011	San Juan	67,750,927	0	100%
1 Apr – 30 Jun 2011	Juneau	68,366,419	0	100%
1 Apr – 30 Jun 2011	Global	405,519,707	0	100%

# 5 Accuracy Standard

**Positioning Accuracy:** The statistical difference, at a 95% probability, between position measurements and a surveyed benchmark for any point within the service volume over any 24-hour interval.

• Horizontal Positioning Accuracy: The statistical difference, at a 95% probability, between horizontal position measurements and a surveyed benchmark for any point within the service volume over any 24-hour interval.

• **Vertical Positioning Accuracy**: The statistical difference, at a 95% probability, between vertical position measurements and a surveyed benchmark for any point within the service volume over any 24-hour interval.

Position/Time Accuracy	Conditions and Constraints
<ul> <li>Global Average Position Domain Accuracy</li> <li>≤ 9m 95% Horizontal Error</li> <li>≤ 15m 95% Vertical Error</li> </ul>	<ul> <li>Defined for a position/time solution meeting the representative user conditions</li> <li>Standard based on a measurement interval of 24 hours averaged over all points in the service volume.</li> </ul>
Worst Site Position Domain Accuracy	• Defined for a position/time solution meeting the
	representative user conditions
• ≤ 17m 95% Horizontal Error	• Standard based on a measurement interval of 24 hours
• $\leq$ 37m 95% Vertical Error	averaged over all points in the service volume.
Time Transfer Domain Accuracy	• Defined for a time transfer solution meeting the
	representative user conditions
• $\leq 40$ nanoseconds time transfer error 95% of time	• Standard based on a measurement interval of 24 hours
(SIS only)	averaged over all points in the service volume.

User Range Accuracy	Conditions and Constraints
Single Frequency C/A-Code	• For any healthy SPS SIS
• $\leq$ 7.8m 9%% Global Average URE during normal	• Neglecting single-frequency ionospheric delay model
operations over All AODs	errors
• $\leq$ 6.0m 95% Global Average URE during operations at	• Including group delay time correction (T <sub>GD</sub> ) errors at
Zero AOD	L1
• ≤ 12.8m 95% Global Average URE during normal	• Including inter-signal bias (P(Y)-code to C/A-code)
operations at Any AOD	errors at L1
Single-Frequency C/A-Code:	• For any healthy SPS SIS
	• Neglecting all perceived pseudorange rate errors
• $\leq$ 6 mm/sec 95% Global Average URRE over any 3-	attributable to pseudorange step changes caused by NAV
second interval during normal operations at Any AOD	message data cutovers
	• Neglecting single-frequency ionospheric delay model
	errors
Single-Frequency C/A-Code:	• For any healthy SPS SIS
	• Neglecting all perceived pseudorange rate errors
• $\leq 2 \text{ mm/sec}^2 95\%$ Global average URA E over any 3-	attributable to pseudorange step changes caused by NAV
second interval during normal operations at Any AOD	message data cutovers
	• Neglecting single-frequency ionospheric delay model
Coordinated Universal Time Offact Funan Accuracy	errors Conditions and Constraints
Coordinated Universal Time Offset Error Accuracy	
• $\leq$ 40 nanoseconds 95% Global average UTCOE	• For any healthy SPS SIS
during normal operations at Any AOD.	

### 5.1 **Position Accuracy**

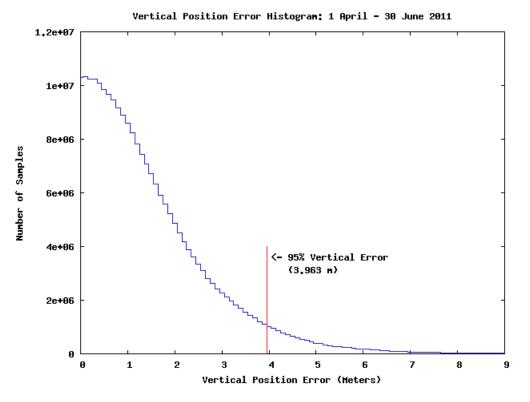
The data used for this section was collected for every second from 1 April through 30 June 2011 at the selected WAAS locations. Table 5-1 provides the 95% and 99.99% horizontal and vertical error accuracies for the quarter. Every twenty-four hour analysis period this quarter passed both the worst-case and global position accuracy requirements set forth by the SPS specification.

Site	95%	95%	99.99%	99.99%
	Horizontal	Vertical	Horizontal	Vertical
	(Meters)	(Meters)	(Meters)	(Meters)
Al bu quer que	2.510	3.797	5.135	9.546
Anchorage	2.567	3.476	4.478	7.747
Atlanta	2.442	3.869	5.040	8.834
Barrow	2.600	3.834	4.522	8.303
Bethel	2.468	3.667	4.690	7.649
Billings	2.038	3.512	4.884	8.828
Boston	2.126	3.479	4.141	8.341
Cleveland	1.971	3.544	5.827	7.813
Cold Bay	2.198	4.052	4.687	7.460
Fairbanks	2.781	3.521	4.555	7.504
Gander	2.161	3.299	4.720	8.926
Honolulu	5.731	6.387	11.761	19.394
Houston	3.057	4.093	6.409	9.755
Iqaluit	2.135	3.481	4.277	10.663
Juneau	2.461	3.277	4.547	7.419
Kansas City	2.037	3.756	4.097	8.846
Kotzebue	2.728	3.585	4.569	7.363
Los Angeles	2.997	4.091	6.871	10.252
Merida	4.218	5.156	8.080	11.849
Mi ami	3.336	4.398	6.441	9.503
Minneapolis	1.958	3.485	6.232	8.642
Oakland	2.852	4.109	6.645	10.451
Salt Lake City	2.204	3.799	5.306	10.159
San Jose Del Cabo	4.418	4.783	8.753	13.462
San Juan	3.607	5.132	8.413	14.449
Seattle	2.077	3.621	4.253	9.624
Tapachul a	5.533	7.218	11.366	17.522
Washington, DC	2.149	3.798	4.275	7.824

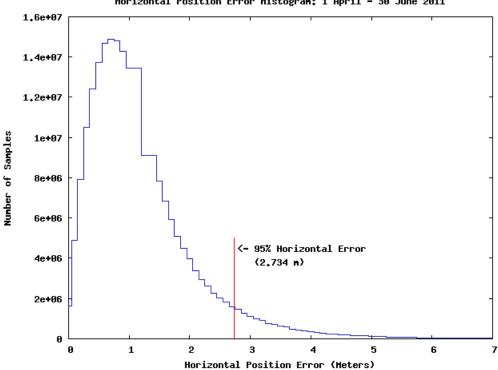
#### Table 5-1 Horizontal & Vertical Accuracy Statistics for the Quarter

Figures 5-1 and 5-2 are the combined histograms of the vertical and horizontal errors for all twenty-eight WAAS sites from 1 April to 30 June 2011.

#### Figure 5-1 Global Vertical Error Histogram



#### Figure 5-2 Global Horizontal Error Histogram

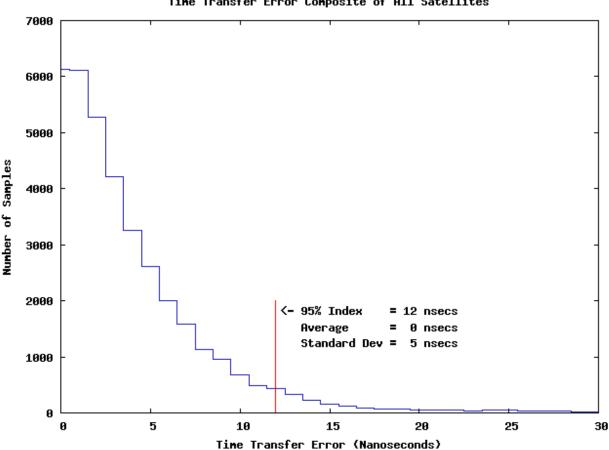


Horizontal Position Error Histogram: 1 April - 30 June 2011

#### 5.2 **Time Transfer Accuracy**

The GPS time error data between 1 April and 30 June 2011 was down loaded from USNO Internet site. The USNO data file contains the time difference between the USNO master clock and GPS system time for each GPS satellites during the time period. Over 10,000 samples of GPS time error are contained in the USNO data file. In order to evaluate the GPS time transfer error, the data file was used to create a histogram (Fig 5-3) to represent the distribution of GPS time error. The histogram was created by taking the absolute value of time difference between the USNO master clock and GPS system time, then creating data bins with one nanosecond precision. The number of samples in each bin was then plotted to form the histogram in Fig 5-3. The mean, standard deviation, and 95% index are within the requirements of GPS SPS time error.

#### Figure 5-3 Time Transfer Error



#### Time Transfer Error Composite of All Satellites

### 5.3 Range Domain Accuracy

Tables 5-3 through 5-5 provide the statistical data for the range error, range rate error and the range acceleration error for each satellite. This data was collected between 1 April and 30 June 2011. A weighted average filter was used for the calculation of the range rate error and the range acceleration error. All Range Domain SPS specifications were met.

#### Table 5-2 Range Error Statistics

#### (Meters)

PRN	RMS Range Error (≤6 m)	Range Error Mean	1σ	95% Range Error	Max Range Error (SPS Spec. ≤ 30 m)	Samples
2	1.517	0.072	1.363	2.953	15.130	14367375
3	1.836	0.211	1.429	3.360	14.431	12406048
4	1.741	-0.546	1.447	3.272	16.142	14029708
5	1.569	-0.613	1.325	2.900	14.397	13977217
6	1.728	0.292	1.350	3.172	13.793	13066980
7	2.120	-1.009	1.451	3.750	12.399	12485307
8	2.200	-0.251	1.613	3.978	11.405	13018223
9	1.985	-0.274	1.517	3.584	17.597	13090165
10	1.886	0.285	1.466	3.499	13.839	12037130
11	1.623	-0.069	1.342	3.011	12.366	12281543
12	1.783	-0.557	1.519	3.544	14.930	14459843
13	1.781	-0.582	1.418	3.335	13.167	13918380
14	1.639	-0.055	1.385	3.176	13.126	14240665
15	1.599	-0.583	1.313	2.899	15.290	13071735
16	1.700	0.036	1.426	3.232	12.998	13594888
17	2.003	-0.653	1.644	3.821	16.873	14359592
18	1.410	0.378	1.141	2.663	14.439	12890281
19	1.734	0.462	1.391	3.275	19.360	12650553
20	1.620	0.141	1.410	3.109	14.066	14225959
21	1.391	0.389	1.102	2.655	13.366	12324542
22	1.809	0.815	1.225	3.260	18.255	12408884
23	1.718	-0.263	1.348	3.160	13.427	12630799
24	1.907	-0.044	1.621	3.562	13.273	13231896
25	1.719	-0.101	1.486	3.357	13.557	13520644
26	1.813	-0.285	1.461	3.401	18.244	12698059
27	2.037	-0.027	1.621	3.757	18.734	13807427
28	2.145	-0.309	1.574	3.848	14.811	12851927
29	1.569	-0.197	1.337	3.045	17.786	13439431
30	2.073	-0.095	1.591	3.850	13.490	6008292
31	1.828	-0.605	1.392	3.378	16.827	14149702
32	1.713	0.589	1.389	3.160	13.289	14276512

#### Table 5-3 Range Rate Error Statistics

#### (Millimeters/ Second)

PRN	Range Rate Error RMS	95% Range Rate Error	Max Range Rate Error	Samples
2	1.516	2.890	139.440	14367375
3	1.823	2.946	153.160	12406048
4	1.477	2.798	125.170	14029708
5	1.468	2.779	155.000	13977217
6	1.782	2.602	162.560	13066980
7	1.492	2.820	153.750	12485307
8	1.840	3.118	156.980	13018223
9	2.009	3.051	156.890	13090165
10	1.789	2.940	175.670	12037130
11	1.500	2.795	125.840	12281543
12	1.616	3.078	152.580	14459843
13	1.568	2.882	162.740	13918380
14	1.610	2.959	153.450	14240665
15	1.489	2.819	145.480	13071735
16	1.534	2.874	148.150	13594888
17	1.679	2.901	140.410	14359592
18	1.453	2.742	154.490	12890281
19	1.463	2.767	128.370	12650553
20	1.541	2.876	165.230	14225959
21	1.538	2.882	145.230	12324542
22	1.596	2.820	154.800	12408884
23	1.463	2.715	133.910	12630799
24	1.931	2.960	192.490	13231896
25	1.443	2.721	152.260	13520644
26	1.473	2.837	132.240	12698059
27	2.264	3.073	191.580	13807427
28	1.608	2.807	133.850	12851927
29	1.559	2.862	152.120	13439431
30	2.298	3.182	253.710	6008292
31	1.586	2.831	258.620	14149702
32	1.527	2.663	169.480	14276512

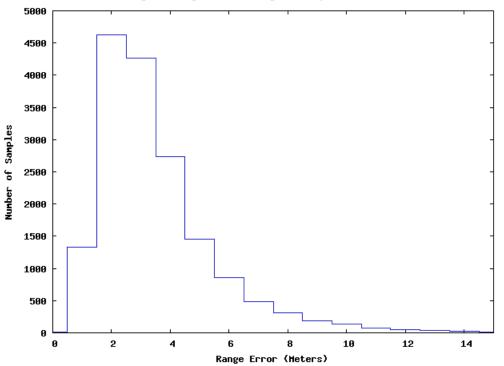
#### Table 5-4 Range Acceleration Error Statistics

#### (Micrometers/Second<sup>2</sup>)

PRN	Range Acceleration	95% Range	Max Range	Samples
	Error RMS (µm/s <sup>2</sup> )	Acceleration Error $(\mu m/s^2)$	Acceleration Error $(\mu m/s^2)$	
2	<u>(µm/s)</u> 10.566	21.392	(µm/s) 1420	14367375
2	13.969	23.058	1420	12406048
3	10.576	20.465	1240	14029708
4	10.610	20.465	1550	13977217
5	14.487	20.951	1630	13066980
6	10.674	20.951	1550	12485307
7				
8	13.242	22.954	1560	13018223
9	15.111	22.339	1560	13090165
10	13.282	22.337	1740	12037130
11	10.981	21.468	1270	12281543
12	10.691	22.142	1520	14459843
13	11.306	21.713	1640	13918380
14	11.379	21.590	1540	14240665
15	10.592	21.564	1470	13071735
16	10.903	22.271	1500	13594888
17	12.276	21.238	1400	14359592
18	10.667	21.634	1560	12890281
19	10.551	21.384	1280	12650553
20	11.088	21.386	1660	14225959
21	10.793	22.022	1460	12324542
22	11.908	21.524	1490	12408884
23	10.810	20.577	1350	12630799
24	14.551	21.681	1890	13231896
25	10.786	20.122	1510	13520644
26	10.543	21.092	1330	12698059
27	17.692	21.747	1910	13807427
28	12.012	21.527	1340	12851927
29	11.212	20.841	1520	13439431
30	17.689	23.159	2540	6008292
31	11.623	21.170	2540	14149702
32	11.583	20.196	1700	14276512

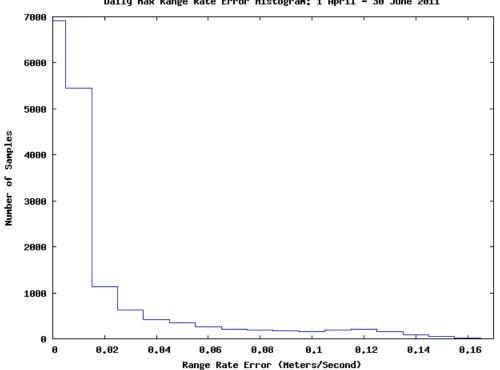
Figures 5-4, 5-5 and 5-6 are graphical representations of the distributions of the maximum range error, range rate error and range acceleration error for all satellites. The highest maximum range error occurred on satellite 22 with an error of 22.116 meters. Satellite 5 had the lowest maximum range error of 10.422 meters.

#### Figure 5-4 Distribution of Daily Max Range Errors

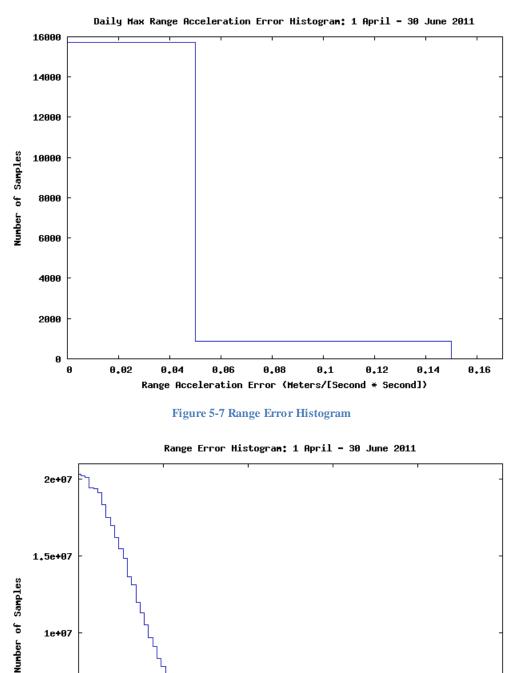


Daily Max Range Error Histogram: 1 April - 30 June 2011





Daily Max Range Rate Error Histogram: 1 April - 30 June 2011



<- 95% Range Error

Range Error (Meters)

6

8

(3,312 m)

4

# Figure 5-6 Distribution of Daily max Range Acceleration Errors

5e+06

0

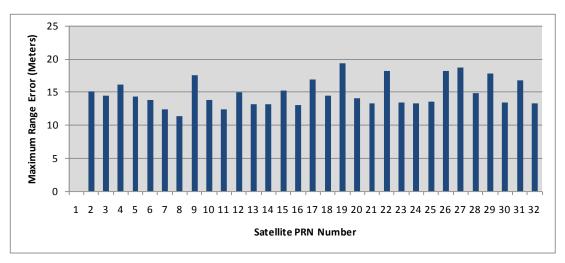
0

2

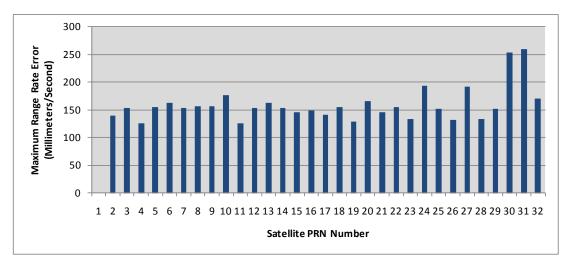
10

July 31, 2011

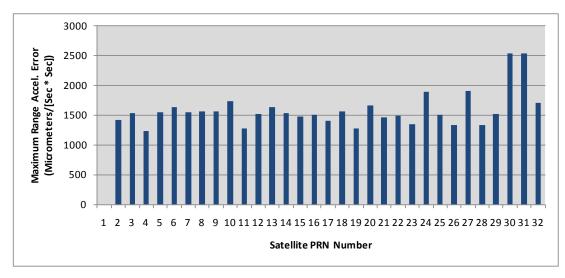








#### Figure 5-10 Maximum Range Acceleration Error Per Satellite



### 6 Solar Storms

Solar storm activity is being monitored in order to assess the possible impact on GPS SPS performance. Solar activity is reported by the Space Environment Center (SEC), a division of the National Oceanic and Atmospheric Administration (NOAA). When storm activity is indicated, ionospheric delays of the GPS signal, satellite outages, position accuracy and availability will be analyzed.

The following article was taken from the SEC web site http://sec.noaa.gov. It briefly explains some of the ideas behind the association of the aurora with geomagnetic activity and a bit about how the 'K-index' or 'K-factor' works.

The aurora is caused by the interaction of high-energy particles (usually electrons) with neutral atoms in the earth's upper atmosphere. These high-energy particles can 'excite' (by collisions) valence electrons that are bound to the neutral atom. The 'excited' electron can then 'de-excite' and return back to its initial, lower energy state, but in the process it releases a photon (a light particle). The combined effect of many photons being released from many atoms results in the aurora display that you see.

The details of how high energy particles are generated during geomagnetic storms constitute an entire discipline of space science in its own right. The basic idea, however, is that the Earth's magnetic field (let us say the 'geomagnetic field') is responding to an outwardly propagating disturbance from the Sun. As the geomagnetic field adjusts to this disturbance, various components of the Earth's field change form, releasing magnetic energy and thereby accelerating charged particles to high energies. These particles, being charged, are forced to stream along the geomagnetic field lines. Some end up in the upper part of the earth's neutral atmosphere and the auroral mechanism begins.

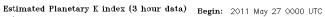
An instrument called a magnetometer may also measure the disturbance of the geomagnetic field. At NOAA's operations center magnetometer data is received from dozens of observatories in one-minute intervals. The data is received at or near to 'real-time' and allows NOAA to keep track of the current state of the geomagnetic conditions. In order to reduce the amount of data NOAA converts the magnetometer data into three-hourly indices, which give a quantitative, but less detailed measure of the level of geomagnetic activity. The K-index scale has a range from 0 to 9 and is directly related to the maximum amount of fluctuation (relative to a quiet day) in the geomagnetic field over a three-hour interval.

The K-index is therefore updated every three hours. The K-index is also necessarily tied to a specific geomagnetic observatory. For locations where there are no observatories, one can only estimate what the local K-index would be by looking at data from the nearest observatory, but this would be subject to some errors from time to time because geomagnetic activity is not always spatially homogenous.

Another item of interest is that the location of the aurora usually changes geomagnetic latitude as the intensity of the geomagnetic storm changes. The location of the aurora often takes on an 'oval-like' shape and is appropriately called the auroral oval.

Figures 6-1 through 6-3 show the K-index for three time periods with significant solar activity. Although there were other days with increased solar activity, these time periods were selected as examples. (See Appendix B for the actual geomagnetic data for this reporting period.)

#### Figure 6-1 K-Index for 27-29 May 2011



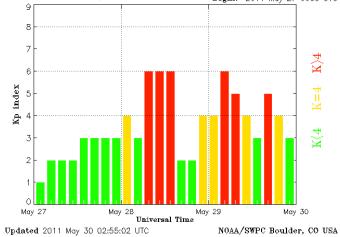
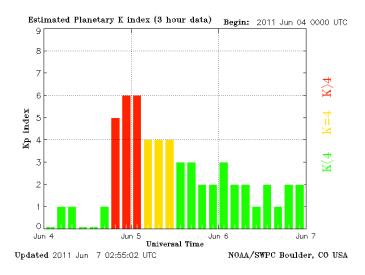
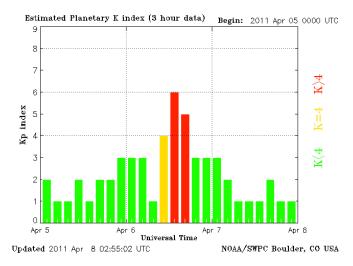


Figure 6-2 K-Index for 4-6 June 2011







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Table 6-1 shows the position accuracy information for the day corresponding to Figure 6-1. The GPS SPS performance met all requirements during all storms that occurred during this quarter.

Site	95%	95%	Maximum	Maximum
	Horizontal	Vertical	Horizontal	Vertical
	(Meters)	(Meters)	(Meters)	(Meters)
Al bu quer que	1.940	5.440	2.340	6.300
Anchorage	2.880	2.970	4.070	5.040
Atlanta	2.680	5.810	3.200	8.480
Barrow	2.670	3.990	3.280	5.140
Bethel	2.390	3.050	3.020	4.960
Billings	3.060	4.260	4.040	5.220
Boston	2.810	4.130	3.270	7.810
Cleveland	2.860	4.490	3.330	7.180
Cold Bay	1.730	3.260	2.230	4.500
Fairbanks	2.900	3.280	6.570	6.840
Gander	3.340	3.810	4.010	5.160
Honolulu	3.130	6.700	3.760	7.860
Houston	2.540	5.410	3.050	6.900
Iqaluit	1.910	4.050	2.600	7.460
Juneau	3.000	3.610	4.020	5.210
Kansas City	2.510	4.960	3.460	6.350
Kotzebue	2.940	2.940	3.570	5.220
Los Angeles	2.900	4.750	4.070	5.420
Meri da	3.720	6.570	4.840	8.290
Miami	3.370	7.480	4.260	8.230
Minneapolis	3.380	4.380	4.320	7.140
Oakland	2.310	4.360	3.640	5.590
Salt Lake City	2.180	4.830	3.240	5.680
San Jose Del Cabo	3.190	5.190	3.470	8.580
San Juan	3.750	4.910	4.360	6.390
Seattle	2.390	4.350	4.070	6.180
Tapachula	Not	Available	Site	Down
Washington, DC	2.880	4.740	3.610	8.260

#### Table 6-1 Horizontal & Vertical Accuracy Statistics for May 28, 2011

# 7 IGS Data

GPS SPS accuracy performance was evaluated at a selection of high rate IGS stations<sup>(1)</sup>. The IGS is a voluntary federation of many world wide agencies that pool resources and permanent GNSS station data to generate precise GNSS products.

High data rate (1 Hz) sites that had high availability in 2006, were outside of the WAAS service area, and provided a good geographic distribution have been selected. To facilitate differentiating between GPS accuracy issues and receiver tracking problems, an automatic data screening function excluded errors greater than 500 meters and or times when VDOP or HDOP were greater than 10. The remaining receiver tracking issues are still included in the processing and are forced into the 50.1 meter histogram bin and are believed to influence the outliers in the 99.99% statistics. The MATE site had a large ramping error on day 267 that appears to be a receiver clock failure. The MATE data for this day has been removed from the statistics computation and trend lines, see figure 7-4.

The Klobachar ionospheric correction model parameters in the global broadcast RINEX navigation data files from the cddis.gsfc.nas/gps/data ftp site were corrupted and caused large daily errors for the equatorial locations. The data was re-processed using Klobachar parameters obtained from the FAA NSTB network or receivers. High quality navigation data is created by voting across all available IGS high rate navigation data. The IGS global navigation data file is not used because it contains occasional errors. (Round off precision, false track records, truncated numbers, probable bit errors in the parent subframe data, and missing updates)

Table 7.1 and Figure 7-1 show the IGS site information and locations. Table 7.2 shows the GPS SPS Accuracy Performance observed at a selection of High Rate IGS sites. Figure 7-2 shows the 95% horizontal accuracy trends at these sites. Figure 7-3 shows the 95% vertical accuracy trends at these sites. A value of zero indicates no data. Figure 7.4 is a position accuracy plots for a MATE site, which are the outliers in the 95% horizontal error trend plots. This example plot suggests that the receiver is encountering hardware resets and tracking problems.

(1) J.M. Dow, R.E. Neilan, G. Gendt, "The International GPS Service (IGS): Celebrating the 10th Anniversary and Looking to the Next Decade," Adv. Space Res. 36 vol. 36, no. 3, pp. 320-326, 2005. Doi: 10.1016/j.asr.2005.05.125

ID	City	Country
GLPS	Puerto Ayora	Ecuador
GUAM	Dededo	Guam
IISC	Bangalore	India
KIRU	Kiruna	Sweden
KOUR	Kourou	French Guyana
MADR	Robledo	Spain
MAL2	Malindi	Kenya
MAS1	Maspalomas	Spain
MOBN	Obninsk	<b>Russian Federation</b>
NNOR	New Norcia	Australia
NRIL	Norilsk	<b>Russian Federation</b>
PETS	Petropavlovsk-Kamchatka	<b>Russian Federation</b>
POL2	Bishkek	Kyrgyzstan
SANT	Santiago	Chile
SUTM	Sutherland	South Africa
TIDB	Tidbinbilla	Australia
USUD	Usuda	Japan

#### Table 7-1 Selected IGS Site Information

#### Figure 7-1 Selected IGS Site Locations

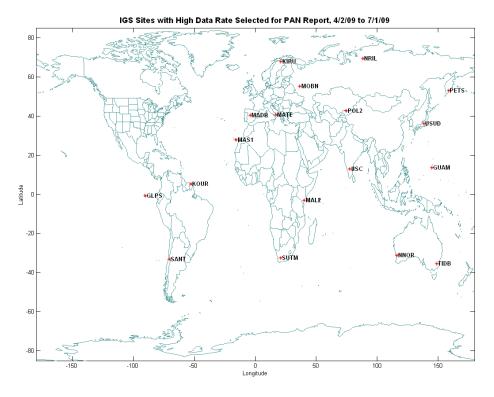
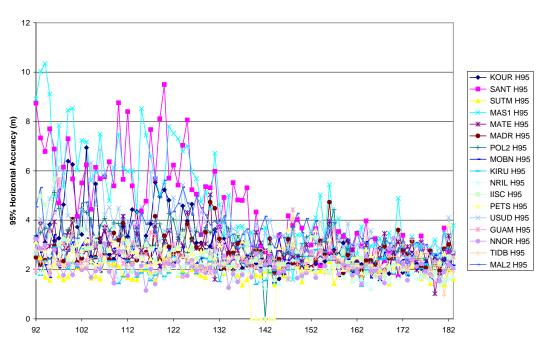


Table 7-2 GPS SP	S Performance	at Selected	High	Rate I	GS Sites
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Site	95%	95%	99.99%	99.99%	Percent
	Horizontal	Vertical	Horizontal	Vertical	Data
	Error (m)	Error (m)	Error (m)	Error (m)	Available
GLPS	n/a	n/a	n/a	n/a	0.00%
GUAM	2.67	7.04	8.72	24.61	99.00%
IISC	2.68	6.68	13.90	19.48	98.90%
KIRU	2.31	4.11	4.39	12.12	98.50%
KOUR	3.32	4.63	7.40	12.51	100.00%
MADR	2.85	4.09	7.42	12.60	99.14%
MAL2	3.86	4.97	7.03	14.40	95.95%
MAS1	6.00	5.89	13.34	15.04	99.99%
MATE	2.88	4.08	7.14	36.61	95.25%
MOBN	2.47	4.32	7.21	11.00	98.83%
NNOR	2.16	4.33	4.05	10.48	99.99%
NRIL	2.07	4.16	5.20	12.46	98.64%
PETS	2.64	5.07	6.83	12.30	92.68%
POL2	3.12	5.05	16.46	24.32	71.52%
SANT	5.01	5.23	13.29	16.05	100.00%
SUTM	1.97	4.34	6.01	10.87	99.22%
TIDB	2.37	4.27	4.29	12.96	98.65%
US UD	3.62	4.53	12.09	11.39	100.00%

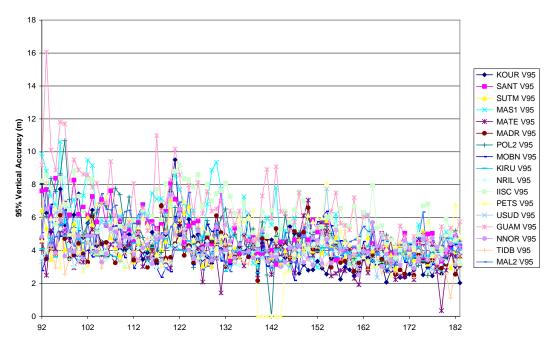




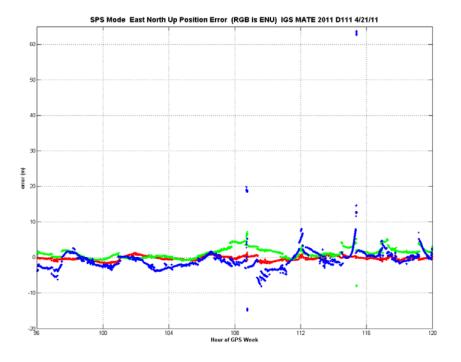
4/2/11 to 7/2/11 95% Horizontal Accuracy Trends

Figure 7-3 GPS SPS 95% Vertical Accuracy Trends at Selected IGS Sites





### Figure 7-4 Example of MATE Receiver 99.99% Outlier



# 8 GPS Test NOTAMs Summary

**GPS test NOTAM:** <u>Global Positioning System test Notices to Airmen</u> - GPS test NOTAMs are issued in the event that GPS is predicted to be unreliable and/or unavailable at a defined location for specific times, as indicated in the NOTAM, due to scheduled testing events.

Status and Problem Reporting	Conditions and Constraints
<ul> <li>Scheduled event affecting service</li> <li>Appropriate GPS Test NOTAM issued to the FAA at least 5 hours prior to the event</li> </ul>	• For any SPS SIS

# 8.1 GPS Test NOTAMs Issued

GPS test NOTAMs were tracked and trended from GPS test NOTAMs posted on the FAA PilotWeb website (https://pilotweb.nas.faa.gov/PilotWeb/). During this reporting period, April 1 through June 30, 2011, there were a total of 20 GPS test NOTAMs issued. The total number of days affected in this reporting period was 65. Tables 8.1 and 8.2 below list the statistics of areas affected and durations. Note that the durations are on a per GPS test NOTAM basis.

#### Table 8-1 GPS test NOTAM Durations

Cumulative duration	727 hours
Minimum duration	3.00 hour
Average duration	36.35 hours
Maximum duration	293.00 hours

#### Table 8-2 GPS Test NOTAM Affected Areas (Square Miles) by Altitude

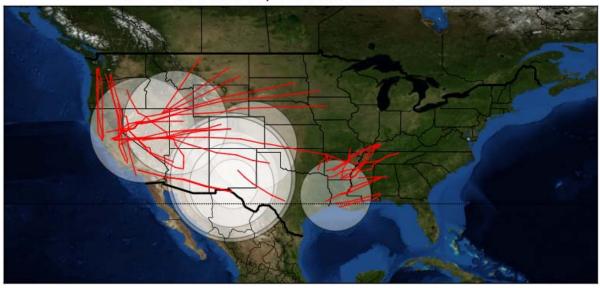
	40,000 feet	25,000 feet	10,000 feet	4,000 feet
Minimum	0	0	0	4
Average	464,374	365,763	200,415	198,249
Maximum	998,910	823,861	509,648	466,900

# 8.2 Tracking and Trending of GPS Test NOTAMs

GPS test NOTAMS have not been tracked and statistically analyzed in a manner for reporting and trending purposes for affects on GPS availability. The GPS test NOTAMs that are tracked and trended for this reporting period were done with a specialized software analysis tool that is designed to not only trend but also archive GPS test NOTAMs. It is designed to trend archived GPS test NOTAMs for any specified time frame. In addition to the data provided in this report, this tool will provide all affected RNAV routes and procedures in a web interface format. The four plots

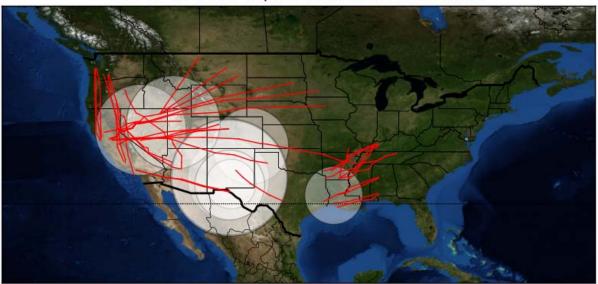
below illustrate a visual depiction of the affected areas at their corresponding altitudes. Note that some GPS test NOTAMs occupy the same area and position but differ in effective dates and/or durations.

## Figure 8-1 GPS Test NOTAMs @ FL400



GPS NOTAMs and impacted RNAV Routes at FL400

#### Figure 8-2 GPS NOTAMs @ FL250



GPS NOTAMs and impacted RNAV Routes at FL250

## Figure 8-3 GPS NOTAMs @ 10k Feet

# GPS NOTAMs and impacted RNAV Routes at 10,000



## Figure 8-4 GPS NOTAMs @ 4k Feet

GPS NOTAMs and impacted RNAV Routes at 4,000



## 8.3 GPS Availability

The impacts to GPS availability are listed below for the corresponding locations and times. The percentage impact to GPS availability indicates that GPS is impacted for X % of the total time that the GPS Test NOTAM is active within the indicated area, centered at the indicated latitude/longitude. The radius column indicates the distance from the latitude/longitude for which the impacted GPS availability extends. Note that the radius listed is for an altitude of 40,000 feet. The impact to GPS availability at lower altitudes is the same. Each row of the following table represents one GPS Test NOTAM. The remaining tables each represent one GPS Test NOTAM.

DATE	TIME	Location(lat/lon)	Radius (nautical miles)	Impact to GPS availability (%)
April 5 -7	0200 - 0800	33.1127N/106.3447W	365	25.00
April 8	19:00 - 22:00	37.3001/103.5621W	340	12.50
April 12- 14	03:00 - 11:00	33.3228/106.2313W	490	33.33
April 15 – 17	02:00 - 12:00	32.4916N/106.0047W	189	33.33
April 18	03:00 - 11:00	33.3228N/106.2313W	490	33.33
April 20 – 23	02:00 - 12:00	32.4916N/106.0047W	189	33.33
May 2 – 5	17:00 - 20:00	40.2602N/113.2936W	395	12.5
May 9	13:00 - 22:00	38.1542N/76.2602W	15	37.5
May 9-12	17:00 - 20:00	40.2602N/113.2936W	395	12.5
May 16 – 20	13:00 - 22:00	38.152N/76.2602W	15	37.5
May 17 – 29	00:00 - 17:00	31.0535N/93.0350W	270	70.83
May 18 – 20	06:00 - 06:00	33.5543N/118.2323W	.33	100
May 23 – 27	13:00 - 22:00	38.1542N/76.2602W	15	37.5
May 25 – 26	19:00 - 23:00	32.3652N/106.1722W	320	16.67
June 27 – 29	19:00 - 23:00	32.3652N/106.1722W	320	16.67

#### Table 8-3 NOTAM Impact to GPS Availability

DATE	TIME	Location (lat/lon)	Radius (nautical miles)	Impact to GPS availability (%)
April 5	16:00 - 23:59	39.3316N/117.4400W	400	33.33
April 6	00:00 - 13:00	39.3316N/117.4400W	400	54.17
April 7	00:00 - 08:15 and 16:00 - 23:50	39.3316N/117.4400W	400	62.50

## Table 8-4 Summary of GPS Test NOTAM 1

## Table 8-5 Summary of GPS Test NOTAM 2

DATE	TIME	Location (lat/lon)	Radius (nautical miles)	Impact to GPS availability (%)
April 12	13:30 - 23:59	39.3316N/117.4400W	400	27.08
April 13	00:00 - 08:15	39.3316N/117.4400W	400	34.38
April 14	00:00 - 07:30	39.3316N/117.4400W	400	31.25

## Table 8-6 Summary of GPS Test NOTAM 3

DATE	TIME	Location (lat/lon)	Radius (nautical miles)	Impact to GPS availability (%)
April 28 - 29	17:00 – 20:00 and 22:00 – 23:59	40.2602N/113.2936W	395	20.83
April 30	17:00 - 20:00	40.2602N/113.2936W	395	12.50

DATE	TIME	Location (lat/lon)	Radius (nautical miles)	Impact to GPS availability (%)
May 23	13:00 - 17:00	32.3652N/106.1722W	320	16.67
May 24	19:00 - 23:00	32.3652N/106.1722W	320	16.67
May 25 - 26	02:00 - 12:00 and 19:00 - 23:00	32.3652N/106.1722W	320	50.00
May 27	02:00 - 12:00	32.3652N/106.1722W	320	33.33
May 30	13:00 - 17:00	32.3652N/106.1722W	320	16.67
May 31	19:00 - 23:00	32.3652N/106.1722W	320	16.67
June 1 – 2	07:01 - 12:00 and 19:00 - 23:00	32.3652N/106.1722W	320	37.50
June 3	07:01 - 12:00	32.3652N/106.1722W	320	20.83
June 6	13:00 - 17:00	32.3652N/106.1722W	320	16.67
June 7	19:00 - 23:00	32.3652N/106.1722W	320	16.67
June 8 – 9	02:00 - 07:00 and 19:00 - 23:00	32.3652N/106.1722W	320	37.5
June 10	02:00 - 07:00	32.3652N/106.1722W	320	20.83
June 13	13:00 - 17:00	32.3652N/106.1722W	320	16.67
June 14	19:00 - 23:00	32.3652N/106.1722W	320	16.67
June 15 – 16	07:01 - 12:00 and 19:00 - 23:00	32.3652N/106.1722W	320	37.50
June 17	07:01 - 12:00	32.3652N/106.1722W	320	20.83
June 20	13:00 - 17:00	32.3652N/106.1722W	320	16.67
June 21	19:00 - 23:00	32.3652N/106.1722W	320	16.67
June 22 – 23	02:00 - 07:00 and 19:00 - 23:00	32.3652N/106.1722W	320	37.5
June 24	02:00 - 07:00	32.3652N/106.1722W	320	20.83
June 27	13:00 - 17:00	32.3652N/106.1722W	320	16.67
June 28	19:00 - 23:00	32.3652N/106.1722W	320	16.67
June 29 – 30	07:01 – 12:00 and 19:00 – 23:00	32.3652N/106.1722W	320	37.50

## Table 8-7 Summary of GPS Test NOTAM 4

DATE	TIME	Location (lat/lon)	Radius (nautical miles)	Impact to GPS availability (%)
June 22 - 23	07:01 - 12:00	33.2933N/106.2457W	460	20.83
June 24	02:00 - 07:00	33.2933N/106.2457W	460	20.83
June 25	02:00 - 12:00	33.2933N/106.2457W	460	33.33

## Table 8-8 Summary of GPS Test NOTAM 5

# 9.1 Appendix A: Performance Summary

User Range Error Accuracy	Conditions and Constraints	Measured
		<b>Performance</b>
Single Frequency C/A-Code	- E	
	• For any healthy SPS SIS	< 2 212
• $\leq$ 7.8m 95% Global Average URE	• Neglecting single-frequency ionospheric delay model errors	≤ 3.313 m
during normal operations over All AODs	<ul> <li>Including group delay time correction (T<sub>GD</sub>)</li> </ul>	
	errors at L1	N/A
• $\leq$ 6.0m 95% Global Average URE during operations at Zero AOD	<ul> <li>Including inter-signal bias (P(Y)-code to C/A-</li> </ul>	$\mathbf{N}/\mathbf{A}$
• $\leq$ 12.8m 95% Global Average	code) errors at L1	
URE during normal operations at		N/A
Any AOD		1 1/ 1 1
Single Frequency C/A-Code	• For any healthy SPS SIS.	
	<ul> <li>Neglecting single-frequency ionospheric delay</li> </ul>	
• $\leq$ 30m 99.94% Global Average	model errors	
URE during normal operations	• Including group delay time correction (T <sub>GD</sub> )	100% Global
	errors at L1	
• $\leq$ 30m 99.79% Worst Case single	• Including inter-signal bias (P(Y)-code to C/A-	
point average during normal	code) errors at L1	100% W CP
operations.	• Standard based on measurement interval of	
	one year; average of daily values within service	
	volume	
	• Standard based on 3 service failures per year,	
	lasting no more than 6 hours each	
User Range Rate	Conditions and Constraints	
Error Accuracy Single-Frequency C/A-Code:	• For any healthy SPS SIS	
Single-Frequency C/A-Code:	<ul> <li>For any healthy SFS SIS</li> <li>Neglecting all perceived pseudorange rate</li> </ul>	
a Communication (Statistication of the statistication of the stati	errors attributable to pseudorange step changes	< 2 192
• ≤ 6 mm/sec 95% Global Average URRE over any 3-second interval	caused by NAV message data cutovers	$\leq$ 3.182 mm/sec
during normal operations at Any	<ul> <li>Neglecting single-frequency ionospheric delay</li> </ul>	
AOD	model errors	
User Range Acceleration	Conditions and Constraints	
Error Accuracy		
Single-Frequency C/A-Code:	• For any healthy SPS SIS	
	• Neglecting all perceived pseudorange rate	
• $\leq 2 \text{ mm/sec}^2 95\%$ Global average	errors attributable to pseudorange step changes	$\leq$ 23.159 mm/s <sup>2</sup>
URAE over any 3-second interval	caused by NAV message data cutovers	
during normal operations at Any	• Neglecting single-frequency ionospheric delay	
AOD	model errors	

#### Table 9-1 Performance Summary

Status and Problem Reporting	Conditions and Constraints	Measured Performance
Scheduled event affecting service		
• Appropriate NANU issued to the	• For any SPS SIS	$\geq$ 132.883 hours
Coast Guard and the FAA at least 48		Prior to event
hours prior to the event		
Unscheduled outage or problem affecting service	• For any SPS SIS	
Appropriate NANU issued to the		$\leq$ 2.1 hours
Coast Guard and the FAA as soon as		$\leq 2.1$ hours
possible after the event		
Operational Satellite Count	Conditions and Constraints	
• $\geq 0.95$ Probability that the	• Applies to the total number of operational	
constellation will have at least 24	satellites in the constellation (averaged over any	
operational satellites regardless of	day); where any satellite which appears in the	
whether those operational satellites	transmitted navigation message almanac is defined	100%
are located in slots or not	to be an operation satellite regardless of whether	
	that satellite is currently broadcasting a healthy SPS	
	SIS or not and regardless of whether the broadcast	
	SPS SIS also satisfies the other performance	
	standards in the SPS performance standard or not.	
PDOP Availability	Conditions and Constraints	100
• $\geq$ 98% global PDOP of 6 or less	• Defined for a position/time solution meeting the	100 %
	representative user conditions and operating within	100.0/
• $\geq$ 88% worst site PDOP of 6 or	the service volume over any 24-hour interval	100 %
less	Conditions and Constraints	
Service Availability	Conditions and Constraints     17m Horizontal (SIS only) 95% threshold	
• ≥ 99% Horizontal Service Availability, average location	<ul> <li>37m Vertical (SIS only) 95% threshold</li> </ul>	100% Horizontal
Availability, average location	<ul> <li>Defined for a position/time solution meeting the</li> </ul>	10070 110112011141
• $\geq$ 99% Vertical Service	representative user conditions and operating within	100% Vertical
Availability, average location	the service volume over any 24-hour interval.	100% vertical
• $\geq$ 90% Horizontal Service	• 17m Horizontal (SIS only) 95% threshold	
Availability, worst-case location	• 37m Vertical (SIS only) 95% threshold	100% Horizontal
Availability, worst-case location	<ul> <li>Defined for a position/time solution meeting the</li> </ul>	100% Horizontai
• $\geq$ 90% Vertical Service	representative user conditions and operating within	100% Vertical
Availability, worst-case location	the service volume over any 24-hour interval.	
Position/Time Accuracy	Conditions and Constraints	
Global Average Position Domain	• Defined for a position/time solution meeting the	
Accuracy	representative user conditions	≤2.734 m Horizontal
	• Standard based on a measurement interval of 24	
• ≤ 9m 95% Horizontal Error	hours averaged over all points in the service	$\leq$ 3.963 m Vertical
• $\leq 15m 95\%$ Vertical Error	volume.	
Worst Site Position Domain	• Defined for a position/time solution meeting the	
Accuracy	representative user conditions	≤ 5.731 m Horizontal
	• Standard based on a measurement interval of 24	
• $\leq 17m 95\%$ Horizontal Error	hours averaged over all points in the service	$\leq$ 7.218 m Vertical
• $\leq 37m 95\%$ Vertical Error	volume.	
Time Transfer Domain Accuracy	• Defined for a time transfer solution meeting the	
	representative user conditions	
• $\leq$ 40 nanoseconds time transfer	• Standard based on a measurement interval of 24	$\leq$ 12 nanoseconds
error 95% of time	hours averaged over all points in the service	
(SIS only)	volume.	

Per-Slot Availability	Conditions and Constraints	
<ul> <li>≥ 0.957 Probability that a slot in the baseline 24-slot configuration will be occupied by a satellite broadcasting a healthy SPS SIS</li> <li>≥ 0.957 Probability that a slot in the expanded configuration will be occupied by a pair of satellites each broadcasting a healthy SPS SIS</li> </ul>	<ul> <li>Calculated as an average over all slots in the 24-slot constellation, normalized annually</li> <li>Applies to satellites broadcasting a healthy SPS SIS that also satisfy the other performance standards in the SPS performance standard.</li> </ul>	Quarter 4 Report On ly
Constellation Availability	Conditions and Constraints	
• $\geq 0.98$ Probability that at least 21 slots out of the 24 will be occupied either by a satellite broadcasting a healthy SPS SIS in the baseline 24- slot configuration or by a pair of satellites each broadcasting a healthy SPS SIS in the expanded slot configuration • $\geq 0.99999$ Probability that at least 20 slots out of the 24 will be occupied either by a satellite broadcasting a healthy SPS SIS in the baseline 24-slot configuration or by a pair of satellites each broadcasting a healthy SPS SIS in the expanded slot configuration	<ul> <li>Calculated as an average over all slots in the 24-slot constellation, normalized annually.</li> <li>Applies to satellites broadcasting a healthy SPS SIS that also satisfies the other performance standards in the SPS performance standard.</li> </ul>	Quarter 4 Report On ly

# 9.2 Appendix B: Geomagnetic Data

Prepared by the U.S. Dept. of Commerce, NOAA, Space Weather Prediction Center

Current Quarter Daily Geomagnetic Data

	Middle Latitude - Fredericksburg -	High Latitude College	Estimated Planetary
Date 2011 04 01 2011 04 02 2011 04 03 2011 04 04 2011 04 05 2011 04 05 2011 04 06 2011 04 07 2011 04 07 2011 04 09 2011 04 10 2011 04 10 2011 04 11 2011 04 12 2011 04 13 2011 04 14 2011 04 15 2011 04 15 2011 04 15 2011 04 16 2011 04 17 2011 04 18 2011 04 18 2011 04 19 2011 04 20 2011 04 21 2011 04 21 2011 04 22 2011 04 23 2011 04 23 2011 04 24 2011 04 25 2011 04 26 2011 04 27 2011 04 28 2011 04 28 2011 04 29 2011 04 30 2011 05 01 2011 05 03 2011 05 03	- Fredericksburg - $A K-indices$ 9 1 1 1 2 2 3 3 3 4 2 3 3 3 4 2 2 2 2 3 3 3 1 3 4 2 2 2 2 2 3 3 1 1 3 3 2 2 2 2 2 2 3 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4 2 2 2 1 4	AK-indicesA14001353421326434632242024433544331663223110084113001274132166653262211000106132125223813334432010031001221939126665222327346542121452321001172010011211016101100116110162011001106610031101011622200110666<	Planetary          K-indices         1       1       2       3       4       3         4       3       3       3       3       4       3         4       3       3       3       3       4       3         3       2       2       1       2       2       1         2       1       2       2       2       1       1         2       1       2       2       2       3       3         3       3       1       4       6       5       3       3         3       1       1       2       1       1       2       1       1         2       0       1       3       2       3       3       3       3       3       3         3       3       1       2       3 <td< td=""></td<>
2011       05       04         2011       05       05         2011       05       06         2011       05       07         2011       05       07         2011       05       08         2011       05       10         2011       05       11         2011       05       12         2011       05       13         2011       05       14         2011       05       15	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$
2011       05       16         2011       05       17         2011       05       18         2011       05       19         2011       05       20         2011       05       21	10       2       2       3       3       3       2       2         8       1       2       3       2       2       2       2         4       1       2       1       1       1       1       1         2       1       0       1       1       1       1       1         2       1       0       1       1       1       1       0         2       1       0       0       1       1       0       0         4       0       0       1       1       2       1       2       2	21       3       2       5       4       5       2       2       2       12         17       1       2       5       5       3       2       1       2       10         5       3       2       1       1       1       2       1       0         1       0       0       1       2       0       0       0       4         1       1       1       0       0       0       0       3       3         1       0       0       1       0       0       0       1       4	2       2       3       3       4       3       2       3         2       2       3       3       2       3       3       3         2       2       1       1       2       1       2       1       2         1       0       1       1       1       0       1       1       1         1       1       0       0       0       1       1       1       0         0       1       1       0       1       1       1       3

2011       05       23       4       1       0       0       1       2       2       5       2       0       0       1       3       0       2       4       1       0       0       1       1       2       2       1       1       3       1       1       1       1       2       2       1       1       3       1 <th>2011 05 22</th> <th>3</th> <th>1 1</th> <th></th> <th></th> <th>1 1</th> <th></th> <th></th> <th>3</th> <th>220</th> <th>0</th> <th>0</th> <th>1</th> <th>1 1</th> <th>4</th> <th>2</th> <th>2 0</th> <th>0</th> <th>1 1</th> <th>1 1</th> <th></th>	2011 05 22	3	1 1			1 1			3	220	0	0	1	1 1	4	2	2 0	0	1 1	1 1	
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# 9.3 Appendix C: Performance Analysis (PAN) Problem Report

In 1993, the FAA began monitoring and analyzing Global Positioning System (GPS) Standard Positioning Service (SPS) performance data. At present, the FAA has approved GPS for IFR and is developing WAAS and LAAS, both of which are GPS augmentation systems. In order to ensure the safe and effective use of GPS and its augmentation systems within the NAS, it is critical that characteristics of GPS performance as well as specific causes for service outages be monitored and understood. To accomplish this objective, GPS SPS performance data is documented in a quarterly GPS Performance Analysis (PAN) report. The PAN report contains data collected at various National Satellite Test Bed (NSTB) and Wide Area Augmentation System (WAAS) reference station locations. This PAN Problem Report will be issued only when the performance data fails to meet the GPS Standard Positioning Service (SPS) Signal Specification.

#### Problem Description:

There were no problems to report for the quarter.

# 9.4 Appendix D: Glossary

The terms and definitions discussed below are taken from the Standard Positioning Service Performance Specification (October 2001). An understanding of these terms and definitions is a necessary prerequisite to full understanding of the Signal Specification.

#### General Terms and Definitions

Almanac Longitude of the Ascending Node (.o): Equatorial angle from the Prime Meridian (Greenwich) at the weekly epoch to the ascending node at the ephemeris reference epoch.

Coarse/Acquisition (C/A) Code: A PRN code sequence used to modulate the GPS L1 carrier.

Corrected Longitude of Ascending Node ( $\Omega k$ ) and Geographic Longitude of the Ascending Node (GLAN): Equatorial angle from the Prime Meridian (Greenwich) to the ascending node, both at arbitrary time T<sub>k</sub>.

**Dilution of Precision (DOP):** The magnifying effect on GPS position error induced by mapping GPS ranging errors into position within the specified coordinate system through the geometry of the position solution. The DOP varies as a function of satellite positions relative to user position. The DOP may be represented in any user local coordinate desired. Examples are HDOP for local horizontal, VDOP for local vertical, PDOP for all three coordinates, and TDOP for time.

Equatorial Angle: An angle along the equator in the direction of Earth rotation.

Geometric Range: The difference between the estimated locations of a GPS satellite and an SPS receiver.

Ground track Equatorial Crossing (GEC,  $\lambda$ , 2 SOPS GLAN): Equatorial angle from the Prime Meridian (Greenwich) to the location a ground track intersects the equator when crossing from the Southern to the Northern hemisphere. GEC is equal to  $\Omega k$  when the argument of latitude  $\Phi$ ) is zero.

**Instantaneous User Range Error (URE):** The difference between the pseudo range measured at a given location and the expected pseudo range, as derived from the navigation message and the true user position, neglecting the bias in receiver clock relative to GPS time. A signal-in-space (SIS) URE includes residual orbit, satellite clock, and group delay errors. A system URE (sometimes known as a User Equivalent Range Error, or UERE) contains all lineof-sight error sources, to include SIS, single-frequency ionosphere model error, troposphere model error, multipath and receiver noise.

**Longitude of Ascending Node (LAN):** A general term for the location of the ascending node – the point that an orbit intersects the equator when crossing from the Southern to the Northern hemisphere.

**Longitude of the Ground track Equatorial Crossing (GEC, \lambda, 2 SOPS GLAN):** Equatorial angle from the Prime Meridian (Greenwich) to the location a ground track intersects the equator when crossing from the Southern to the Northern hemisphere. GEC is equal to  $\Omega$ k when the argument of latitude $\Phi$ ) is zero.

Mean Down Time (MDT): A measure of time required to restore function after any downing event.

Mean Time Between Downing Events (MTBDE): A measure of time between any downing events.

Mean Time Between Failures (MTBF): A measure of time between unscheduled downing events.

Mean Time to Restore (MTTR): A measure of time required to restore function after an unscheduled downing event.

**Navigation Message:** Data contained in each satellite's ranging signal and consisting of the ranging signal time-of-transmission, the transmitting satellite's orbital elements, an almanac containing abbreviated orbital element

information to support satellite selection, ranging measurement correction information, and status flags. The message structure is described in Section 2.1.2 of the SPS Performance Standard.

**Operational Satellite:** A GPS satellite which is capable of, but is not necessarily transmitting a usable ranging signal.

**PDOP Availability:** Defined to be the percentage of time over any 24-hour interval that the PDOP value is less than or equal to its threshold for any point within the service volume.

**Positioning Accuracy:** Defined to be the statistical difference, at a 95% probability, between position measurements and a surveyed benchmark for any point within the service volume over any 24-hour interval.

• Horizontal Positioning Accuracy: Defined to be the statistical difference, at a 95% probability, between horizontal position measurements and a surveyed benchmark for any point within the service volume over any 24-hour interval.

• Vertical Positioning Accuracy: Defined to be the statistical difference, at a 95% probability, between vertical position measurements and a surveyed benchmark for any point within the service volume over any 24-hour interval.

**Position Solution:** An estimate of a user's location derived from ranging signal measurements and navigation data from GPS.

**Position Solution Geometry:** The set of direction cosines that define the instantaneous relationship of each satellite's ranging signal vector to each of the position solution coordinate axes.

**Pseudo Random Noise (PRN):** A binary sequence that appears to be random over a specified time interval unless the shift register configuration and initial conditions for generating the sequence are known. Each satellite generates a unique PRN sequence that is effectively uncorrelated (orthogonal) to any other satellite's code over the integration time constant of a receiver's code tracking loop.

**Representative SPS Receiver:** The minimum signal reception and processing assumptions employed by the U.S. Government to characterize SPS performance in accordance with performance standards defined in Section 3 of the SPS Performance Standard. Representative SPS receiver capability assumptions are identified in Section 2.2 of the SPS Performance Standard.

**Right Ascension of Ascending Node (RAAN):** Equatorial angle from the celestial principal direction to the ascending node.

**Root Mean Square (RMS) SIS URE:** A statistic that represents instantaneous SIS URE performance in an RMS sense over some sample interval. The statistic can be for an individual satellite or for the entire constellation. The sample interval for URE assessment used in the SPS Performance Standard is 24 hours.

**Selective Availability:** Protection technique formerly employed to deny full system accuracy to unauthorized users. SA was discontinued effective midnight May 1, 2000.

**Service Availability:** Defined to be the percentage of time over any 24-hour interval that the predicted 95% positioning error is less than its threshold for any given point within the service volume.

• Horizontal Service Availability: Defined to be the percentage of time over any 24-hour interval that the predicted 95% horizontal error is less than its threshold for any point within the service volume.

• Vertical Service Availability: Defined to be the percentage of time over any 24-hour interval that the predicted 95% vertical error is less than its threshold for any point within the service volume.

Service Degradation: A condition over a time interval during which one or more SPS performance standards are not supported.

**Service Failure:** A condition over a time interval during which a healthy GPS satellite's ranging signal exceeds the Not-to-Exceed (NTE) SPS SIS URE tolerance.

**Service Reliability:** The percentage of time over a specified time interval that the instantaneous SIS SPS URE is maintained within a specified reliability threshold at any given point within the service volume, for all healthy GPS satellites.

**Service Volume:** The spatial volume supported by SPS performance standards. Specifically, the SPS Performance Standard supports the terrestrial service volume. The terrestrial service volume covers from the surface of the Earth up to an altitude of 3,000 kilometers.

SPS Performance Envelope: The range of nominal variation in specified aspects of SPS performance.

**SPS Performance Standard:** A quantifiable minimum level for a specified aspect of GPS SPS performance. SPS performance standards are defined in Section 3.0.

**SPS Ranging Signal:** An electromagnetic signal originating from an operational satellite. The SPS ranging signal consists of a Pseudo Random Noise (PRN) C/A code, a timing reference and sufficient data to support the position solution generation process. A description of the GPS SPS signal is provided in Section 2. The formal definition of the SPS ranging signal is provided in ICDGPS-200C.

**SPS Ranging Signal Measurement:** The difference between the ranging signal time of reception (as determined by the receiver's clock) and the time of transmission derived from the navigation signal (as defined by the satellite's clock) multiplied by the speed of light. Also known as the *pseudo range*.

#### SPS SIS User Range Error (URE) Statistic:

• A satellite SPS SIS URE statistic is defined to be the Root Mean Square (RMS) difference between SPS ranging signal measurements (neglecting user clock bias and errors due to propagation environment and receiver), and "true" ranges between the satellite and an SPS user at any point within the service volume over a specified time interval.

• A constellation SPS SIS URE statistic is defined to be the average of all satellite SPS SIS URE statistics over a specified time interval.

**Time Transfer Accuracy Relative to UTC (USNO):** The difference at a 95% probability between user UTC time estimates and UTC (USNO) at any point within the service volume over any 24-hour interval.

Transient Behavior: Short-term behavior not consistent with steady-state expectations.

**Usable SPS Ranging Signal:** An SPS ranging signal that can be received, processed, and used in a position solution by a receiver with representative SPS receiver capabilities.

**User Navigation Error (UNE):** Given a sufficiently stationary and ergodic satellite constellation ranging error behavior over a minimum sample interval, multiplication of the DOP and a constellation ranging error standard deviation value will yield an approximation of the RMS position error. This RMS approximation is known as the UNE (UHNE for horizontal, UVNE for vertical, and so on). The user is cautioned that any divergence away from the stationary and ergodic assumptions will cause the UNE to diverge from a RMS value based on actual measurements.

User Range Accuracy (URA): A conservative representation of each satellite's expected  $(1\sigma)$  SIS URE performance (excluding residual group delay) based on historical data. A URA value is provided that is representative over the curve fit interval of the navigation data from which the URA is read. The URA is a coarse representation of the URE statistic in that it is quantized to levels represented in ICDGPS200C.